

Documentation for:

YED/A429/R1L2/VF8 (Lattice FPGA)

ARINC 429 to 8-Relay converter



YED Avionics Limited

Park House 10, Park Street Bristol Avon BS1 5HX UK

Tel: +44 (0) 117 907 4761

E-mail: support@yed.com Web: www.yed.com © YED Avionics Limited Version 1.0

Documentation for: YED/A429/R1/DAT3-5V Converter

Amendment Page

Version	Date of Issue	Change Reference	Remarks
1.0	30 Apr 2015	N/A	Original Issue

Original Author: G.S. Brownett

Date of Origin: 30th April 2015

File Ref: A429-R1L2-VF8-Lattice.man.doc

Firmware Version: V1.0

CONDITIONS GOVERNING THIS DOCUMENTATION

1. Purpose of document

The information for the YED/A429-R1L2/VF8 is provided for purposes of supplying interface information for use by the end user.

2. No Grant of Proprietary Rights

Nothing herein shall be construed as granting to the receiving party any proprietary rights or any licence in respect of YED's proprietary information.

3. Confidentiality

No information contained herein shall be used, copied, re-transmitted to third parties or otherwise made use of except for the express purpose defined above in the section covering the 'Purpose of document'.

4. <u>Disclaimer</u>

All information, including illustrations, is believed to be reliable. Users however, should independently evaluate the suitability of the product for their particular application. *YED Avionics Limited* makes no warranties as to the accuracy or completeness of the information, and disclaims any liability regarding its use. *YED's* only obligations are those contained in the Standard Terms and Conditions of Sale for any incidental, indirect or consequential damages arising from the sale, resale, use or misuse of the product. See http://www.yed.com/terms.php for more information.

Information contained in this manual is subject to change without notice.

CONTENTS

1	INTRODUCTION	4
1.1	Firmware	5
1.2	ENVIRONMENTAL, AIRWORTHINESS AND EMC	5
1.3	SPECIFICATION	6
2	CONFIGURATION OF CONVERTER	8
2.1		
2.2		
2.3		
2.4		
2.5		
2.6		
3	CONNECTOR PIN OUT (D15 PLUG)	13
4	MECHANICAL DIMENSIONS FOR THE ENCLOSURE	13
5	TYPICAL INTERCONNECT DRAWING	14
6	INSTALLATION	17
6.1	ELECTRICAL CONSIDERATIONS	17
6.2	Materials not supplied	17
6.3	Mounting considerations.	17
6.4		
6.5		
•	6.5.1 Removal	
_	6.5.2 Replacement	
	Continued Airworthiness	
_		
7	ENVIRONMENTAL & EMC	
Q		20

© YED Avionics Limited

Version 1.0

Documentation for: YED/A429-R1L2/VF8 Converter

1 Introduction

This manual contains specification data, installation and instructions for the YED/A429-R1L2/VF8, ARINC-429 to 8-channel relay converter.

This converter will monitor the individual bits from two selected ARINC 429 Labels and use this information to control the state of up to 8 volt-free relay contacts.

Each relay contact is dependent upon a discrete bit within a user selectable ARINC 429 Label. If the discrete field bit changes state then a corresponding relay contact will switch state also.

Two user selectable ARINC 429 Labels settings are possible via DIP switches mounted on the PCB. Each relay can be mapped to either the first or second Label (A or B) in any combination. eg. Four relays mapped to Label A (1st Label) and four to Label B (2nd Label) **OR** all eight relays mapped to one Label.

SDI and SSM values for each Label are selectable via DIP switches or the values can be set to "Don't care" whereby the SDI and/or SSM values will be ignored.

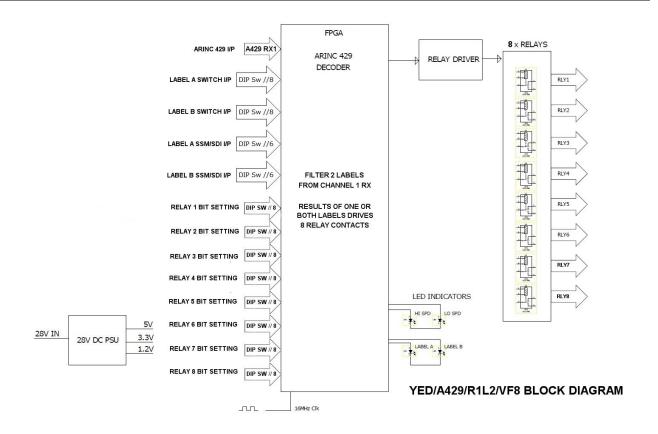
Each relay state can be configured independently to be normally open or normally closed via its respective DIP switch. The relay contacts are latched and will maintain the state of the last received data word until new data is received. Providing both the ARM and Contact of the relay at the I/O connector allows maximum versatility for the OEM designer.

The converter can receive high or low speed ARINC 429 data automatically without user intervention. The PCB contains LED's that indicate whether the received bus is low speed (12.5 kHz) or high speed (100kHz) and whether Label A and/or Label B are being decoded.

The ARINC 429 receiver is coupled to the data bus via an Optocoupler to maintain electrical isolation and EMI immunity.

The unit is powered from an external 28 VDC (18-60V) nominal supply with internal current and thermal (102 °C) fuse. The power inputs are also reverse polarity protected and incorporate all the standard YED EMC/EMI surge protection techniques including a power supply monitoring device, which will cause the system to reset in the event of any problems with the main supply.

A block diagram of the main components of the converter is shown overleaf.



BLOCK DIAGRAM OF CONVERTER INTERNALS

1.1 Firmware

This design is based upon a Field Programmable Gate Array (FPGA) only. There is no microprocessor and therefore RTCA DO-178 certification is not required for this product.

1.2 Environmental, Airworthiness and EMC

The YED/A429-R1L2/VF8 Converter has been designed to meet DO-160D test categories listed later in this manual.

The unit has also been subjected to an Explosive Decompression test from 15,000 feet to 50,000 feet in a period of less than 100mS without effect. See Annex 1.

Version 1.0

Documentation for: YED/A429-R1L2/VF8 Converter

1.3 Specification

The YED/A429-R1L2/VF8 has the following features: -

Physical

The YED/A429-R1L2/VF8 attaches to the airframe via four mounting holes. See paragraph titled "Enclosure Outline Drawing" for further details.

The enclosure is a CNC machined box with Anodised and an Alocrom 1200 finish.

Height	31.0mm
Width	89.0mm
Length	166.0mm

Electrical

Reverse polarity protected.

Indicators (on PCB)

LED bit rate reception: Hi Speed/Low Speed LED Label reception: 3 LEDs. Label A & B.

ARINC 429 Input

Number of receivers...... 1

Input is via opto-coupler.

Bit Rate 12.5kHz or 100kHz. (auto detecting)

Relay outputs

Number of relay outputs..... Relay 1 operates:.... Channel 1 Rx, Label A/B ???, bit # '?' Channel 1 Rx, Label A/B ???, bit # '?' Relay 1 operates:.... Channel 1 Rx, Label A/B ???, bit # '?' Relay 1 operates:.... Channel 1 Rx, Label A/B ???, bit # '?' Relay 1 operates:.... Relay 1 operates:.... Channel 1 Rx, Label A/B ???, bit # '?' Channel 1 Rx, Label A/B ???, bit # '?' Relay 1 operates:.... Relay 1 operates:.... Channel 1 Rx, Label A/B ???, bit # '?' Channel 1 Rx, Label A/B ???, bit # '?' Relay 1 operates:....

Volt Free Relay contact: Rated at 1 amp @ 110Vdc (non-inductive)

Contact resistance: 100mR contact resistance.

© YED Avionics Limited

Version 1.0

Documentation for: YED/A429-R1L2/VF8 Converter

ARING 429 Labels filtered (Any 2 from 256)	
Label A:	User selectable Label A via SMD switches
Label B:	User selectable Label B via SMD switches
ARINC 429 SSM and SDI filtering	
SSM filtering for Label A/B:	11,10,01,00 or XX (Don't care) via SMD switches
SDI filtering for Label A/B:	11,10,01,00 or XX (Don't care) via SMD switches
Connector	
Industry Standard D25 sub-miniature socket.	
Environmental	
Operating temperature range	-10 to +85 degrees C
Operating temperature range	TO LO TOU DESTEES C.

2 Configuration of converter

2.1 Description of setting Labels

The values of each of the Labels are set by the adjusting the surface mounted DIP switches on the PCB as shown below.



2.2 Label switch description

Labels are coded in Octal.
Red spot denotes a switch set to ON.

LAB-L8	LAB-L7	LAB-L6	LAB-L5	LAB-L4	LAB-L3	LAB-L2	LAB-L1	LABEL
ON		ON	ON	ON		ON	ON	273
ON		ON	ON	ON	ON			274

2.3 SDI and SSM switch description

The SDI and SSM switch is coded as shown below.

SDI-10	SDI-09	DON'T CARE	SSM-31	SSM-30	DON'T CARE
			ON	ON	
			ON	ON	
			ON	ON	_

SDI-10 refers to Bit 10 of the ARINC 429 word

SDI-11 refers to Bit 11 of the ARINC 429 word

SDI- Don't care means ignore SDI-10 and SDI-9 settings.

SSM-31 refers to Bit 31 of the ARINC 429 word

SSM-30 refers to Bit 30 of the ARINC 429 word

SSM- Don't care means ignore SSM-31 and SSM-30 settings.

Don't care means that there will not be any filtering of the respective SDI and/or SSM fields and overrides any other setting for the field.

2.4 Discrete bit selection and assignment to relay

There are eight 8-bit DIP switches that are used to select the required mapped bit within the received ARINC 429 words against Label A or Label B. Each of these switches is configured as 5-bits

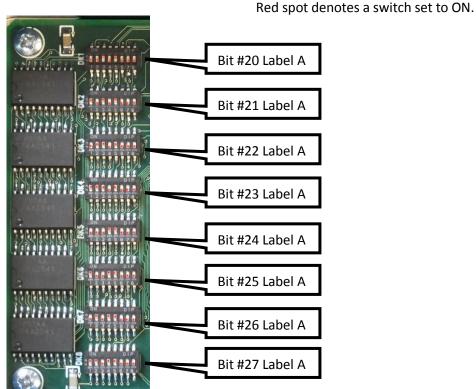
SW1 RB-0	SW2 RB-1	SW3 RB2	SW4 RB3	SW5 RB4	SW6 INV	SW7 A or B	SW8 N/A	BIT #
		ON		ON			Х	20
ON		ON		ON			Х	21
	ON	ON		ON			Х	22
ON	ON	ON		ON			Х	23
			ON	ON			X	24
ON			ON	ON			X	25
	ON		ON	ON			Х	26
ON	ON		ON	ON			Χ	27

SW1 (LSB) thru SW5 (MSB) is used to select the bit number from a range of 0-31.

SW6 when switched ON will invert the relay contact state.

SW7 when OFF selects Label A as the source otherwise Label B when switched ON.

SW8 is not used.



420 Bt /VE

2.5 ARINC 429 Label and data word format

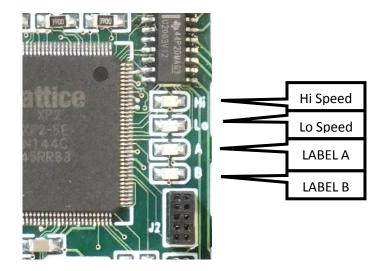
A typical ARINC 429 data word is shown below.

- Starting at Bit-32 is PARITY
- Bits 31 & 30 are the SSM filed, which indicates the status of the data.
- Bit-29 is the sign bit of the data.
- Bits28 thru 9 is the data field for this type of data word. Note that the SDI field on bits 10 & 9 is not present for this particular data word.
- The Label field shown here as 076.

32	31	30	29	28	27	26	25	24	23	22	21	20	19	18	17	16	15	14	13	12	11	10	9	8	7	6	5	4	3	2	1
P	SS [2		S										ALT	ITU	DE												LAI (0				
0	1	1	0	0 MS	0 B	1	0	1	0	0	1	0	0	1	0	0	0	0	1	1	0	0 L	0 SB	0 LSE	1	1	1	1	1	0 N	0 ISB
												- CANADA	21												6	- 00	I.	7	14.		0

2.6 LED indication showing reception of Labels A and B and bit rate

The image below shows the area of the printed circuit board that contains Label reception status and bit rate LED indicators. If LED 'A' or 'B' is lit then the label is being received and decoded. 'Hi' or Lo' speed reception is indicated by the appropriate LED.



3 Connector Pin Out (D15 Plug)

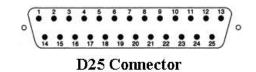
The converter interface is via a single filtered 25-way D-sub connector (Plug). This will accommodate power, ARINC 429 input and contact outputs.

The pin configuration is as follows:

PIN	Description	Comment
1	+28V DC Power	External Power rail input.
2	0V (Gnd)	Ground potential (chassis)
3	A429 Rx1+ ARINC 429 input	ARINC 429 receiver channel 1 (input) +Ve
4	A429 Rx1- ARINC 429 input	ARINC 429 receiver channel 1 (input) –Ve
5	A429 GND	ARINC 429 screen connection. (chassis)
6	Reserved	
7	Relay Contact 1 Arm	RELAY 1 Arm (output) Normally open
8	Relay Contact 1 Contact	RELAY 1 Contact (output) Normally open
9	Relay Contact 2 Arm	RELAY 2 Arm (output) Normally open
10	Relay Contact 2 Contact	RELAY 2 Contact (output) Normally open
11	Relay Contact 3 Arm	RELAY 3 Arm (output) Normally open
12	Relay Contact 3 Contact	RELAY 3 Contact (output) Normally open
13	Relay Contact 4 Arm	RELAY 4 Arm (output) Normally open
14	Relay Contact 4 Contact	RELAY 4 Contact (output) Normally open
15	Relay Contact 5 Arm	RELAY 5 Arm (output) Normally open
16	Relay Contact 5 Contact	RELAY 5 Contact (output) Normally open
17	Relay Contact 6 Arm	RELAY 6 Arm (output) Normally open
18	Relay Contact 6 Contact	RELAY 6 Contact (output) Normally open
19	Relay Contact 7 Arm	RELAY 7 Arm (output) Normally open
20	Relay Contact 7 Contact	RELAY 7 Contact (output) Normally open
21	Relay Contact 8 Arm	RELAY 8 Arm (output) Normally open
22	Relay Contact 8 Contact	RELAY 8 Contact (output) Normally open
23	Reserved	
24	Reserved	
25	Reserved	

Table 2 – J1 Pin Description

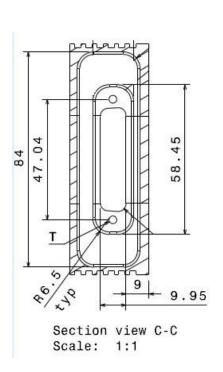


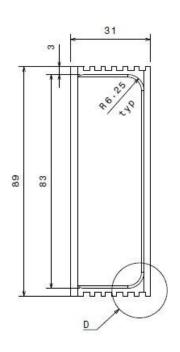


Version 1.0

4 Mechanical Dimensions for the Enclosure

The enclosure is a YED proprietary design. It is a CNC machined aluminium enclosure consisting of a lid and a base section. This enclosure has been designed to meet the most stringent explosive decompression and EMC/EMI requirements. The base of the enclosure contains an EMC gasket that provides an EMC/EMI and environmental seal. There is also the option to have a seal fitted between the 25-way D-sub connector and the inside of the enclosure thus making the unit totally sealed.



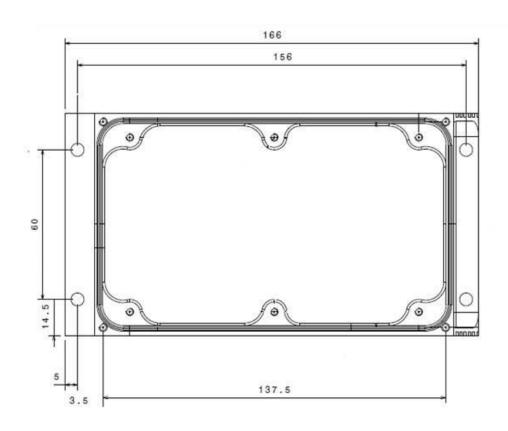




© YED Avionics Limited

Documentation for: YED/A429-R1L2/VF8 Converter

Version 1.0



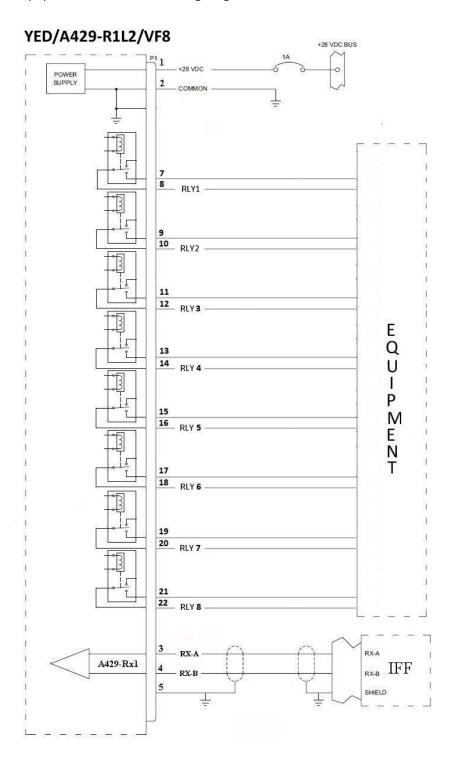
HOLE CODE	TYPE
	Ø5.2MM THRU (TOP AND BTM FLANGE)

MATERIAL A/ALLOY 6028 (HE30TF)



5 Typical interconnect drawing

A typical equipment interconnect wiring diagram is shown below.



6 Installation

This section provides details for the installation of the YED/A429-R1L2/VF8 Converter, including configuration and mounting procedures. Follow the procedures and recommendations found in this section to assure a successful installation.

6.1 Electrical considerations

A circuit breaker such as a Klixon 7277-2-1 or equivalent should be considered for connecting the power from the aircraft supply to this converter – even though the converter is internally fused.

6.2 Materials not supplied

• Wire: MIL-W-22759/16 or equivalent

Shielded wire: MIL-C-27500 or equivalent

• Mounting Screws, 4 each.

6.3 Mounting considerations

The YED/A429-R1L2/VF8 can be mounted in the avionics bay, shelf or other suitable structure. It can be mounted in any orientation.

6.4 Wiring

Use 22 to 24 AWG wire for all connections.

Fabricate wiring harness, and test all wiring for continuity and for shorts. Ensure aircraft power is present on the correct pins of J1; refer to Table 1.

6.5 Removal and replacement

6.5.1 Removal

- 1. Open the circuit breaker powering the YED/A429-R1L2/VF8.
- 2. Remove the connector.
- 3. Remove four (4) screws securing the converter to the airframe.

6.5.2 Replacement

- 1. Open the circuit breaker powering the YED/A429-R1L2/VF8
- 2. Secure the converter to the airframe with four (4) screws.
- 3. Attach the connector and secure
- 4. Close the circuit breaker.
- 5. Perform operational test of the YED/A429-R1L2/VF8

6.6 Continued Airworthiness

6.6.1 Scheduled Maintenance

•	Recommended periodic scheduled servicing	None
•	Recommended periodic scheduled preventative	
	maintenance tests	None
•	Recommended periodic inspections	None
•	Recommended period overhaul period	None
•	Special inspection requirements	None

There are no Airworthiness limitations associated with the installation of this converter.



7 Environmental & EMC

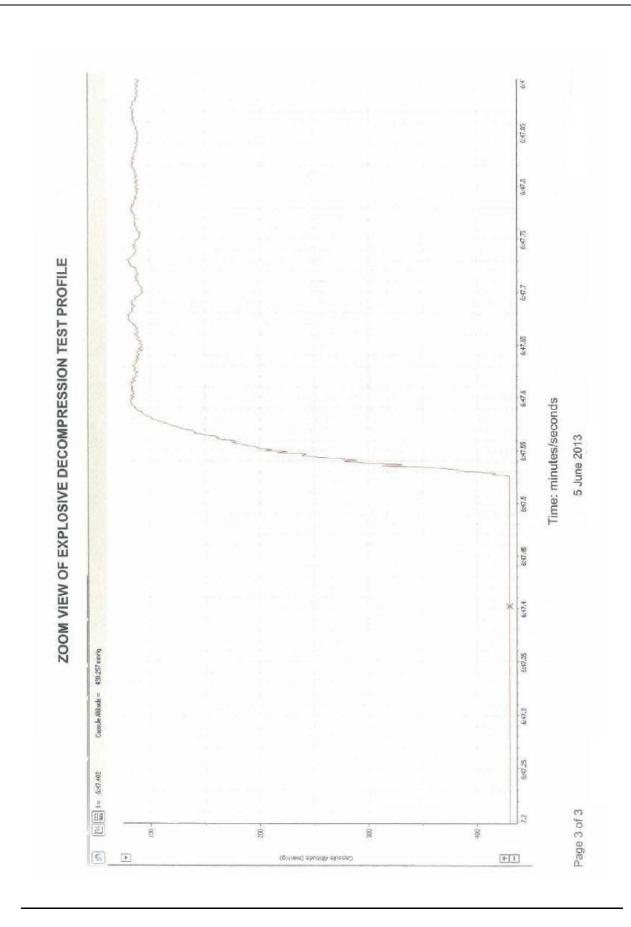
The YED/A429-R1L2/VF8 has been designed to meet the environmental test categories detailed below in accordance with RTCA DO-160D, Environmental Conditions and Test Procedure for Airborne Equipment.

Section	Category	Remarks
4.0 Temperature and Altitude	A1, A2	25,000 feet.
5.0 Temperature and variation	B, C	
6.0 Humidity	A	
7.0 Operational Shock and Crash Safety	В	
8.0 Vibration	C,M	
9.0 Explosion Proofness	X	Not tested – See Annex 1
10.0 Waterproofness	X	Not tested
11.0 Fluids susceptibility	X	Not tested
12.0 Sand and Dust	X	Not tested
13.0 Fungus Resistance	X	Not tested
14.0 Salt Spray	X	Not tested
15.0 Magnetic Effect	Z	
16.0 Power Input	A	
17.0 Voltage Spike	В	
18.0 AF Conducted Susceptibility – Power Inputs	A	
19.0 Induced Signal Susceptibility	A, Z	
20.0 Radio Frequency Susceptibility (Radiated and Conducted)	T, V	
21.0 Emission of Radio Frequency Energy	A, Z	
22.0 Lightening Induced Transient	A,B,Z	Discrete sense pins and ARINC 429
Susceptibility		driver output.
23.0 Lightening Direct Effects	X	Not tested
24.0 Icing	X	Not tested
25.0 ESD	X	Not tested

8 Annex 1 – Explosive Decompression tests

	DECOMPRESSION TESTING OF	NAVIGATION	UNIT
١.	QinetiQ Building 800 Hypobaric Facility (Boscon decompression testing of a Converter unit. The reason for this testing, in conjectable clearance to be given for the equipment to be	unction with other requir	TACAN red testing, is to
2.	The equipment to be tested was supplied	on 22 rd May 2013 and com	nprised:
	- YED TACAN Converter unit.	Local Serial I	No. 001.
3.	Prior to the decompression test, the manufacturer had verified to the supplying the equipment that the equipment was serviceable. The equipment did not require to be functioning during the testing and was to be returned to the manufacturer for examination post testing.		
4.	The decompression test was carried out in the Hypobaric Chamber (Building 800) or May 2013 to the following decompression test profile:		
	 Chamber ascent to 15,000 ft at a rate of 5,000 Hold for a minimum of 1 minute. Rapid decompression to 50,000 ft (To reach 9 time period). Hold chamber altitude for a minimum of 5 minute. Chamber descent to ground level at 5,000 ft/m 	0% of the final altitude wit	hin a 0.1-second
5.	The altitude profile within the chamber was recorded on an ADI Instruments Power Lab data acquisition system. Annex A shows the test profile from; ground level (760 mmHg) to 15,000 to (428 mmHg), explosive decompression to 50,000 ft (87 mmHg) in less than 0.1 second, hold for a minimum of 5 minutes, descent back down to ground level. Chamber ascent from ground level to 15,000 ft and descent from 50,000 ft to ground level was carried out at a rate of 5,000 ft /min. Annex B shows the actual rapid decompression in greater detail during which the chamber reached 90% of the final altitude in a 0.05-second time period. The rates of change of altitude, or pressure, within the chamber are controlled in units of feet per minute (ft/min). For this reason, the pressure recordings show a curvature due to the non-linearity of the altitude/pressure relationship.		
).	The chamber decompression test proceeded without incident and as far as could be seen without any effect on the equipment. The Boscombe Down representative will return the equipment to the manufacturer for examination.		
	This documented detail of the testing has been produced by Chas Taylor (Air Division). (Boscombe Down).		

Page 21



Page 22